

Name:
Enrolment No:



UNIVERSITY OF PETROLEUM AND ENERGY STUDIES

End Semester Examination, December 2018

Programme Name: M. Tech. CFD

Semester : I

Course Name : Introduction to CFD

Time : 03

hrs.

Course Code : ASEG 7001

Max. Marks: 100

Nos. of page(s) : 03

Instructions: Assume any missing data appropriately.

SECTION A

S. No.		Marks	CO
Q 1	List the various physical boundary conditions encountered in a non-isothermal fluid flow.	4	CO1
Q 2	When the law of conservation of mass is applied to finite control volume Ω , moving with flow, the governing equation is given by $\frac{D}{Dt} \iiint \rho d\Omega = 0$ Convert this equation to divergence form, applicable to finite control volume fixed in space.	4	CO1
Q 3	Sketch the various models of fluid flow used for derivation of governing equations. Write down the forms of equations that emanate from these models on applications conservation laws.	4	CO1
Q 4	Illuminate the need of a body fitted coordinate system for the solution of governing flow equations using finite difference method.	4	CO3
Q 5	Compare with illustrations, the explicit and implicit schemes for solution of partial differential equations.	4	CO2

SECTION B

Q 6	Apply the law of conservation of momentum for an infinitesimally small element of a viscous fluid moving in space and hence deduce the momentum equation for fluids in non-conservation form.	10	CO1
Q 7	Consider the following system of equations	10	CO2

$$\frac{\partial u_1}{\partial t} + \frac{\partial u_2}{\partial x} = 0,$$

$$\frac{\partial u_2}{\partial t} + \frac{\partial u_3}{\partial x} = 0,$$

$$\frac{\partial u_3}{\partial t} + 4\frac{\partial u_1}{\partial x} - 17\frac{\partial u_2}{\partial x} + 8\frac{\partial u_3}{\partial x} = 0.$$

Classify this system of equations as hyperbolic or elliptic, based on the eigenvalue method.

Q 8	<p>Analyze the stability of explicit FTCS scheme for solution of transient, one-dimensional heat conduction equation and hence establish the stability criterion. Assume the error to be of the form</p> $\epsilon_m(x, t) = e^{at} e^{ik_m x}$ <p style="text-align: center;">OR</p> <p>An explicit scheme for solving the first-order wave equation is given by:</p> $u_j^{n+1} = u_j^n - c \frac{\Delta t}{\Delta x} \left(\frac{u_{j+1}^{n+1} - u_{j-1}^{n+1}}{2} \right)$ <p>Apply the Fourier stability analysis to this scheme, and determine the stability restrictions, if any.</p>	10	CO4
Q 9	<p>Discuss an explicit time marching algorithm for the solution of transient Euler equations in 2-dimensions.</p>	10	CO2

SECTION-C

Q 10	<p>Consider a two-dimensional square plate ABCD with edges AB and CD maintained at temperatures of 200 °K and 100 °K respectively. The other two edges DA and BC are also maintained at temperatures of 200 °K, except at the corners C and D. Find the steady state temperatures of at least 9 locations on the plate. Take $AB=BC=CD=DA= 4$ cm. Use pure Gauss-Seidel relaxation scheme for at least 4 iterations.</p> <p>The two-dimensional steady state heat conduction is governed by</p> $\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} = 0$ <p style="text-align: center;">OR</p>	20	CO5
-------------	---	-----------	------------

	<p>Consider a 6 cm long steel, whose left and right ends are kept at fixed temperatures of 0 °C and 100 °C respectively. The initial ($t = 0$ s) temperature at other locations on the rod is 25 °C.</p> <p>Predict the temperature at any five locations on the rod after 5 seconds ($t = 5$ s) using an explicit Forward in Time and Central in Space (FTCS) finite difference scheme. Hint: Take $\Delta x = 0.01$ m.</p>		
Q 11	<p>Derive the <i>modified equation</i> that results from the first order forward in time and backward in space discretization of the first order wave equation.</p> $\frac{\partial u}{\partial t} + a \frac{\partial u}{\partial x} = 0$ <p>Discuss the nature of dominating error for the above discretization and suggest means to minimize them.</p>	20	CO4

Name:

Enrolment No:



UNIVERSITY OF PETROLEUM AND ENERGY STUDIES

End Semester Examination, December 2018

Programme Name: M. Tech. CFD

Semester : I

Course Name : Introduction to CFD

Time : 03 hrs.

Course Code : ASEG 7001

Max. Marks: 100

Nos. of page(s) : 03

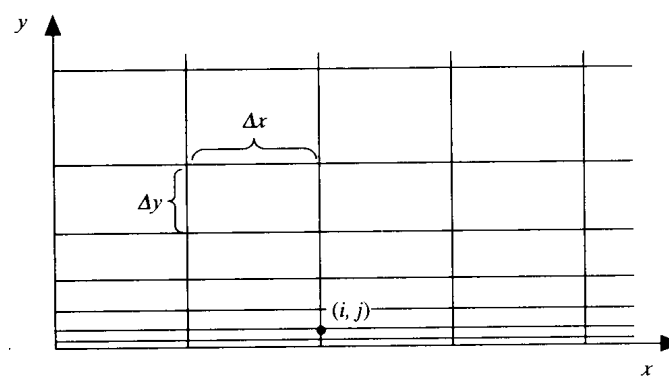
Instructions: Assume any missing data appropriately.

SECTION A

S. No.		Marks	CO										
Q 1	<p>Write down second order accurate finite difference stencils for discretization of the following derivatives.</p> <p>a. $\frac{\partial^2 u}{\partial y^2}$</p> <p>b. $\frac{\partial u}{\partial y}$</p>	4	CO1										
Q 2	<p>Consider the viscous flow of air over a flat plate. At a given station in the flow direction, the variation of the flow velocity, u, in the direction perpendicular to the plate (the y direction) is given at discrete grid points equally spaced in y direction with $\Delta y = 2.54$ mm.</p> <table><thead><tr><th>y (mm)</th><th>u (m/s)</th></tr></thead><tbody><tr><td>0</td><td>0</td></tr><tr><td>2.54</td><td>45.72</td></tr><tr><td>5.08</td><td>87.41</td></tr><tr><td>7.62</td><td>125.0</td></tr></tbody></table> <p>Imagine that the values of u listed above are discrete values at discrete grid points located at $y = 0, 2.54, 5.08$ and 7.62 mm the same nature as would be obtained from a numerical finite difference solution of the flow field. For viscosity coefficient, $\mu = 1.7895 \times 10^{-5}$ kg/m-s, using these discrete values; Calculate the shear stress at the wall τ_w three different ways, namely:</p> <p>a. Using a first order one sided difference</p> <p>b. Using the second order one sided difference</p>	y (mm)	u (m/s)	0	0	2.54	45.72	5.08	87.41	7.62	125.0	4	CO3
y (mm)	u (m/s)												
0	0												
2.54	45.72												
5.08	87.41												
7.62	125.0												
Q 3	Define <i>numerical diffusion and dispersion</i> . Discuss the effect of numerical diffusion	4	CO4										

	and dispersion on the solution of the one-dimensional scalar wave equation using the explicit Forward in Time and Backward in Space (FTBS) scheme. Suggest methods to alleviate the diffusive error.		
Q 4	Find the values of Mach number (M_∞) for which the system of equations given below is hyperbolic. $(1 - M_\infty^2) \frac{\partial u'}{\partial x} + \frac{\partial v'}{\partial y} = 0$ $\frac{\partial u'}{\partial y} - \frac{\partial v'}{\partial x} = 0$	4	CO1
Q 5	Elucidate the need of grid and equation transformation for the solution flow over complex geometries using finite difference method.	4	CO3

SECTION B

Q 6	<p>Consider the numerical solution of steady viscous flow over a flat plate using a finite difference scheme. To calculate the details of this flow near the surface, very fine mesh, stretched in transverse direction is required as shown in figure below.</p>  <p>The solution requires an equispaced Cartesian grid in computational plane (ξ, η) which can be obtained through following direct transformations</p> $\xi = x$ $\eta = \ln(1 + y)$ <p>If the continuity equation for above flow in physical plane (x, y) is $\frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} = 0$, find the continuity equation that is required to be solved in the computational plane.</p>	10	CO3
Q 7	Consider the 2-dimensional transient heat conduction equation given below. The	10	CO2

	<p>Crank-Nicolson discretization of the equation results in a pentadiagonal system of equations. Demonstrate an algorithm to solve the system of equations iteratively.</p> $\frac{\partial T}{\partial t} = \alpha \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right)$		
Q 8	<p>Discuss the solution of Laplace equation in 2-dimensions using an explicit five point Gauss-Seidel Scheme point iterative scheme. Suggest modifications that can accelerate this scheme.</p>	10	CO2
Q 9	<p>Derive a third order, one-sided finite difference discretization of first order derivative $\frac{\partial u}{\partial x}$.</p> <p style="text-align: center;">OR</p> <p>Derive a fourth order accurate finite difference stencil for the mixed derivative $\frac{\partial^2 u}{\partial x^2}$.</p>	10	CO2
SECTION-C			
Q 10	<p>Deduce the <i>modified equation</i> for the solution of the first order wave equation using Lax Method given by</p> $\frac{u_j^{n+1} - (u_{j+1}^n + u_{j-1}^n)/2}{\Delta t} + c \frac{u_{j+1}^n - u_{j-1}^n}{2 \Delta x} = 0$ <p>Hence, discuss the effect of the dominating error on the solution obtained.</p>	20	CO4
Q 11	<p>Consider the Couette flow between two plates, characterized by the parabolic equation,</p> $\frac{\partial u}{\partial t} - \nu \frac{\partial^2 u}{\partial y^2} = 0, \quad \nu = 0.000217 \text{ m}^2/\text{s}$ <p>with initial and boundary conditions given as:</p> <p>Initial conditions at $t = 0$ $\begin{cases} u = u_0 = 40 \text{ m/s}, & y = 0 \\ u = 0, & 0 < y \leq h \end{cases}$</p> <p>Boundary conditions at $t > 0$ $\begin{cases} u = u_o = 40 \text{ m/s}, & y = 0 \\ u = 0, & y = h \end{cases}$</p> <p>Evaluate the velocity at the midway location between the plates, at $t = 0.4$ seconds using Crank-Nicolson Scheme. Take $h = 0.04$ m and use $\Delta y = 0.004$ m.</p>	20	CO5

OR

Consider square plate PQRS whose edges PQ, QR and SP are maintained at temperature of 400 °K whereas the edge RS is maintained at 200 °K. Find the steady state temperatures of at least 9 locations on the plate. Take $PQ=QR=RS=SP= 4$ cm. Use a point iterative relaxation scheme for at least 4 iterations with an over-relaxation factor of 1.2.

The two-dimensional steady state heat conduction is governed by

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} = 0$$